

DPP

DAILY PRACTICE PROBLEMS

Class : XIth
Date :

Solutions

Subject : PHYSICS
DPP No. : 5

Topic :- GRAVITATION

1 (c)

$$\text{Escape velocity } v = \sqrt{\frac{2GM}{R}}$$

If star rotates with angular velocity ω

$$\text{Then } \omega = \frac{v}{R} = \frac{1}{R} \sqrt{\frac{2GM}{R}} = \sqrt{\frac{2GM}{R^3}}$$

2 (d)

Time period (T) of a synchronous satellite around the earth is given by

$$T^2 = \frac{4\pi^2 r^3}{Gm_e} \Rightarrow r = \left(\frac{T^2 Gm_e}{4\pi^2} \right)^{1/3}$$

Substituting the given values, we get

$$r = \left[\frac{(24 \times 60 \times 60)^2 + 6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{4 \times \frac{22}{7} \times \frac{22}{7}} \right]^{1/3}$$

$$r = 42.08 \times 10^6 m$$

$$\therefore \frac{r}{r_e} = \frac{42.08 \times 10^6 m}{6.37 \times 10^6 m} = 6.6 \Rightarrow r = 6.6 r_e$$

3 (d)

$$\text{Kinetic energy of the satellite is } K = \frac{GMm}{2r} \dots(i)$$

$$\text{Potential energy of the satellite is } U = -\frac{GMm}{r} \dots(ii)$$

$$\text{Total energy of the satellite is } E = -\frac{GMm}{2r} \dots(iii)$$

$$\text{Divide (iii) by (i), we get } \frac{E}{K} = -1 \text{ or } E = -K$$

$$\text{Divide (iii) by (ii), we get } \frac{E}{U} = \frac{1}{2} \text{ or } E = \frac{U}{2}$$

6 (b)

$F = Gm_1m_2/r^2$, thus on increasing masses and reducing distance r , force of gravitational attraction F will increase

7 (b)

Time period is independent of mass. Therefore their periods of revolution will be same.

8 (c)

Kinetic energy = Potential energy

$$\frac{1}{2}m(kv_e)^2 = \frac{mgh}{1 + \frac{h}{R}} \Rightarrow \frac{1}{2}mk^2 2gR = \frac{mgh}{1 + \frac{h}{R}} \Rightarrow h = \frac{Rk^2}{1 - k^2}$$

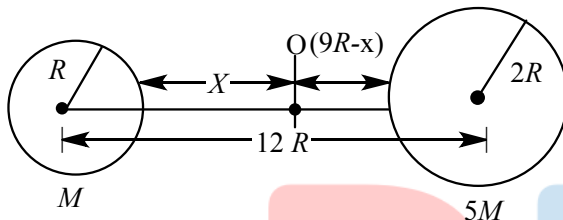
Height of Projectile from the earth's surface = h

Height from the centre $r = R + h = R + \frac{Rk^2}{1 - k^2}$

By solving $r = \frac{R}{1 - k^2}$

9 (c)

Let at O there will be a collision. If smaller sphere moves x distance to reach at O , then bigger sphere will move a distance of $(9R - x)$



$$a_{\text{small}} = \frac{F}{M} = \frac{G \times 5M}{(12R - x)^2}$$

$$a_{\text{big}} = \frac{F}{5M} = \frac{GM}{(12R - x)^2}$$

$$x = \frac{1}{2} a_{\text{small}} t^2 = \frac{1}{2} \frac{G \times 5M}{(12R - x)^2} t^2 \quad \dots(i)$$

$$(9R - x) = \frac{1}{2} a_{\text{big}} t^2 = \frac{1}{2} \frac{GM}{(12R - x)^2} t^2 \quad \dots(ii)$$

Thus, dividing Eq. (i) by Eq. (ii), we get

$$\therefore \frac{x}{9R - x} = 5$$

$$\Rightarrow x = 45R - 5x$$

$$\Rightarrow 6x = 45R$$

$$\Rightarrow x = 7.5 R$$

10 (b)

In circular orbit of a satellite of potential energy

$$= -2 \times (\text{kinetic energy})$$

$$= -2 \times \frac{1}{2} m v^2 = -m v^2$$

Just to escape from the gravitational pull, its total mechanical energy should be zero.

Therefore, its kinetic energy should be $+m v^2$

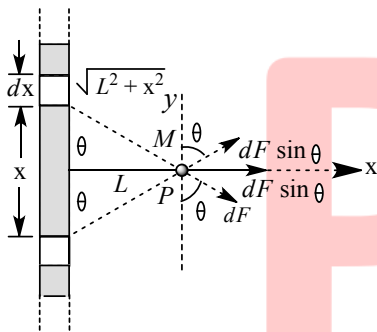
- 11 (a) Acceleration due to gravity at height h is

$$g' = \frac{g}{\left(1 + \frac{h}{R_e}\right)^2} = \frac{g}{\left(1 + \frac{32}{6400}\right)^2} = 0.99 g$$

- 12 (a) $v = \sqrt{2gR}$. If acceleration due to gravity and radius of the planet, both are double that of earth then escape velocity will be two times, i.e. $v_p = 2v_e$

- 13 (b) Potential energy $U = \frac{-GMm}{r} = -\frac{GMm}{R+h}$
 $U_{initial} = -\frac{GMm}{3R}$ and $U_{final} = -\frac{GMm}{2R}$
 Loss in PE = gain in KE = $\frac{GMm}{2R} - \frac{GMm}{3R} = \frac{GMm}{6R}$

- 14 (c)



Let the mass M be placed symmetrically

$$\Rightarrow F_{\text{net}} = \int_{-\infty}^{\infty} dF \sin \theta = \int_{-\infty}^{\infty} \frac{GM(\lambda dx)}{X^2 + L^2} \frac{L}{\sqrt{X^2 + L^2}}$$

$$\Rightarrow F_{\text{net}} = GM\lambda L \int_{-\infty}^{\infty} \frac{dx}{(X^2 + L^2)^{3/2}}$$

$$\Rightarrow F_{\text{net}} = \frac{GM\lambda L}{L^2} (2)$$

$$\Rightarrow F_{\text{net}} = \frac{2GM\lambda}{L^2}$$

- 15 (d) The system will be bound at points where total energy is negative. In the given curve at point A, B and C the P.E. is more than K.E.

- 17 (c)

$$v_e = \sqrt{2gR} \Rightarrow \frac{v_A}{v_B} = \sqrt{\frac{g_A}{g_B} \times \frac{R_A}{R_B}} = \sqrt{x \times r} \therefore \frac{v_A}{v_B} = \sqrt{rx}$$

18 **(b)**

$$g' = g\left(1 - \frac{d}{R}\right) \Rightarrow \frac{g}{n} = g\left(1 - \frac{d}{R}\right) \Rightarrow d = \left(\frac{n-1}{n}\right)R$$

20 **(b)**

$$U_{(r)} = \begin{cases} -\frac{GMm}{r}, r \geq R \\ -\frac{GMm}{R}, r < R \end{cases}$$

PE

ANSWER-KEY										
Q.	1	2	3	4	5	6	7	8	9	10
A.	C	D	D	A	C	B	B	C	C	B
Q.	11	12	13	14	15	16	17	18	19	20
A.	A	A	B	C	D	B	C	B	B	B

P E