

# DPP

DAILY PRACTICE PROBLEMS

Class : XIth  
Date :

Solutions

SUBJECT : PHYSICS  
DPP No. : 4

## Topic :- GRAVITATION

1 (a)

$$\text{Time period, } T = \frac{2\pi R}{\sqrt{\frac{GM_m}{R}}} = \frac{2\pi R^{3/2}}{\sqrt{GM_m}}$$

Where the symbols have their meaning as given in the question

Squaring both sides, we get

$$T^2 = \frac{4\pi^2 R^3}{GM_m}$$

2 (d)

$$\text{Orbital velocity } v_0 = \sqrt{\frac{GM}{r}} = \sqrt{\frac{gR^2}{r}} \text{ and } v_0 = r\omega$$

$$\text{This gives } r^3 = \frac{R^2 g}{\omega^2}$$

3 (d)

$$\text{Escape velocity } v_e = \sqrt{\frac{2GM}{R}}$$

$$v_e = \sqrt{\frac{2G \frac{4}{3}\pi R^3 \times d}{R}}$$

$$\sqrt{2G \frac{4}{3}\pi R^3 \times d} = R \sqrt{\frac{8}{3}\pi Gd}$$

where  $d$  = mean density of earth

$$\therefore v_e \propto R\sqrt{d}$$

$$\therefore \frac{v_e}{v_p} = \frac{R_e}{R_p} \sqrt{\frac{d_e}{d_p}}$$

$$= \frac{R_e}{2R_e} \sqrt{\frac{d_e}{d_e}}$$

$$= v_p = 2v_e$$

$$= 2 \times 11 = 22 \text{ km s}^{-1}$$

5 (d)

Here,  $u = 20 \text{ ms}^{-1}$ ,  $m = 500 \text{ g} = 0.5 \text{ kg}$ ,  $t = 20 \text{ s}$

Using Newton's equation of motion

$$s = ut + \frac{1}{2}gt^2$$

$$0 = 20 \times 20 + \frac{1}{2}(-g)(20)^2$$

or  $g = 2 \text{ ms}^{-2}$

$$\begin{aligned} \therefore \text{Weight of body on planet} &= mg \\ &= 0.5 \times 2 = 1 \text{ N} \end{aligned}$$

8 (c)

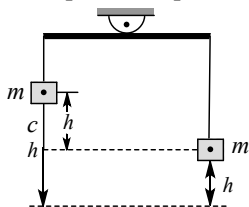
Angular momentum remains constant

$$mv_1d_1 = mv_2d_2 \Rightarrow v_2 = \frac{v_1d_1}{d_2}$$

9 (b)

As with height  $g$  varies as

$$g' = \frac{g}{[1 + h/R]^2} = g \left[ 1 - \frac{2h}{R} \right]$$



and in according with figure  $h_1 > h_2$ , so  $W_1$  will be lesser than  $W_2$  and

$$W_2 - W_1 = mg_2 - mg_1 = 2mg \left[ \frac{h_1}{R} - \frac{h_2}{R} \right]$$

$$\text{or } W_2 - W_1 = 2m \frac{GMh}{R^2}$$

$$\left[ \text{as } g = \frac{GM}{R^2} \text{ and } (h_1 - h_2) = h \right]$$

$$\text{or } W_2 - W_1 = \frac{2m_h G}{R^3} \left( \frac{4}{3} \pi R^3 \rho \right)$$

$$= \frac{8}{3} \pi \rho G m h \left[ \text{as } M = \frac{4}{3} \pi R^3 \rho \right]$$

11 (c)

Time period of nearby satellite

$$\begin{aligned} T &= 2\pi \sqrt{\frac{r^3}{GM}} \\ &= 2\pi \sqrt{\frac{R^3}{GM}} \\ &= \frac{2\pi(R^3)^{1/2}}{\left[ G \frac{4}{3} \pi R^3 \rho \right]^{1/2}} = \sqrt{\frac{3\pi}{G\rho}} \end{aligned}$$

12 (a)

The acceleration due to gravity ( $g$ ) is given by

$$g = \frac{GM}{R^2}$$

where  $M$  is mass,  $G$  the gravitational constant and  $R$  the radius.

Since, planets have a spherical shape

$$V = \frac{4}{3}\pi r^3$$

Also, mass ( $M$ ) = volume( $V$ )  $\times$  density( $\rho$ )

$$g = \frac{G \frac{4}{3}\pi R^3 \rho}{R^2}$$

$$\Rightarrow g = \frac{4G\pi\rho R}{3}$$

Given,  $R_1:R_2 = 2:3$

$$\rho_1:\rho_2 = \frac{3}{2}$$

$$\therefore \frac{g_1}{g_2} = \frac{\rho_1 R_1}{\rho_2 R_2} = \frac{3}{2} \times \frac{2}{3} = 1$$

13

(c)

The maximum velocity with which a body must be projected in the atmosphere, so as to enable it to just overcome the gravitational pull, is known as escape velocity.

Escape velocity from earth's surface is

$$v_{es} = \sqrt{\frac{2GM_e}{R_e}}$$

$$= \sqrt{\frac{2G \cdot \frac{4}{3}\pi R_e^3 d_e}{R_e}} \quad \left( \because M = \frac{4}{3}\pi R_e^3 d_e \right)$$

or  $v_{es} \propto \sqrt{d_e} \times R_e \dots(i)$

similarly, for a planet

$$v'_{es} \propto \sqrt{d_p} \times R_p \dots(ii)$$

So,  $\frac{v_{es}}{v'_{es}} = \left(\frac{d_e}{d_p}\right)^{1/2} \times \frac{R_e}{R_p}$

Given,  $d_p = \frac{1}{4}d_e, R_p = 2R_e$

$$\frac{v_{es}}{v'_{es}} = \left(\frac{d_e}{\frac{1}{4}d_e}\right)^{1/2} \times \frac{R_e}{2R_e}$$

$$= (4)^{1/2} \times \frac{1}{2}$$

$$= 2 \times \frac{1}{2} = 1$$

So,  $\frac{v_{es}}{v'_{es}} = 1:1$

14 (a)

The value of acceleration due to gravity  $g$  at height  $h$  above the surface of earth is

$$g_h = \frac{g}{\left(1 + \frac{h}{R}\right)^2}$$

Where  $R$  is radius of earth.

$$\therefore \frac{g}{g_h} = \left(1 + \frac{h}{R}\right)^2$$

15 (b)

$$v = \sqrt{\frac{GM}{r}}$$

16 (b)

Angular momentum is conserved in central field

17 (d)

The true weight of a body is given by  $mg$  and with height  $g$  decrease

$$\text{So, } \frac{W_s}{W_E} = \frac{mg'}{mg} = \frac{1}{[1 + (h/R)]^2} \left[ \text{as } g' = \frac{g}{[1 + (h/R)]^2} \right]$$

But here,  $h = 7R - R = 6R$ , ie,  $h/R = 6$

$$\text{So, } W_s = \frac{W_E}{(1 + 6)^2} = \frac{10}{49} = 0.2\text{N}$$

18 (d)

$$v_e = \sqrt{\frac{2GM}{(R + h)}}$$

19 (a)

$$g' = g \left( \frac{R}{R + h} \right)^2 = g \left( \frac{R}{R + 2R} \right)^2 = \frac{g}{9}$$

20 (c)

$$\frac{g_m}{g_e} = \frac{G(M/8)}{GM/R_e^2} = \frac{R_e^2}{8R_m^2}; \dots(i)$$

$$\text{Given, } \frac{mg_m}{mg_e} = \frac{1}{6}$$

$$\text{or } \frac{g_m}{g_e} = \frac{1}{6} \dots(ii)$$

$$\text{From Eqs. (i) and (ii); } \frac{R_e^2}{8R_m^2} = \frac{1}{6}$$

$$\text{or } R_e = \sqrt{8/6}R_m$$

ANSWER-KEY										
Q.	1	2	3	4	5	6	7	8	9	10
A.	A	D	D	A	D	B	D	C	B	A
Q.	11	12	13	14	15	16	17	18	19	20
A.	C	A	C	A	B	B	D	D	A	C

P E